

# Wildfire modelling: the CFD approach

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### Aix\*Marseille Some general recommandations in CFD modeling

- Evaluate even coarsly the characteristic length and time scales of the problem
- The mesh size must not be defined arbritray but from physic based considerations
- Simplify as possible the problem (2D/3D, compressible/incompressible flow ...)
- Turbulence, Combustion, Radiation models ?
- Use adapted numerical scheme avoiding numerical dissipation and a certain guaranty of stability (with a flux-limiter strategy, TVD, Ultra-Sharp ...)
- If possible (highly recommanded) compare some numerical results with experimental data
- Evoid the use of the following expression « Our model is validated ... »
- Consider that « All models are wrong but someones are usefull »

### Aix Marseille Universite Physical phenomena governing wildfires



















NESDIS/OSEI NOAA-15 AVHRR HRPT RGB=CH3,CH2,CH1 05/11/2000 01:30 UTC

## Wildfires: a multi-scale, non linear problem (Los Alamos fire May 2000)

140 km

**New Mexico** 

**500 μm** 









### Another mechanism of fire propagation: firebrands



Travelling distance by brands; few kms ! Brands are the main source of vulnerability of houses located in WUI.

## Wildfires: some physical scales

Fire (combustion, turbulence): Flame thickness: d<sub>F</sub> ~ 500 μm

Soot + hot gas Radiation + Convection Fuel: drying, pyrolysis, combustion

H<sub>fuel</sub> Size

ABL (turbulence/canopy): Large scale:  $L_t \sim H_{Fuel}$ Micro-scale:  $\eta \sim 100 - 500 \mu m$ Thermal plume (turbulence)

Morvan, Fire Technology 2011

Radiation (extinction length scale): L<sub>R</sub> ~ 2xH<sub>Fuel</sub> / LAI 0.1 – 5 m



## Wildfires modeling (Weber 1991, Sullivan 2009 ...)

- •Statistical models (empirical),
  - Mc Arthur (1966)  $R=f(f_i)$
- Semi-empirirical models,

Rothermel (1972)  $R = \xi I_r / \rho \Delta h_i$ 

## Physical models (radiative, full physics),

 Albini (1985), De Mestre (1989), Balbi, Santoni & al (1998), Siméoni & al (2001), Chatelon, Rossi, Marcelli & al (2010)
 Grishin (1985), Larini, Morvan, Porterie & al (1996)
 Clark & al (1996), Lin & al (1997),
 Sero-Guillaume & al (2002), Rehm & al (2003), Mandel & al (2004), Mell & al (2005), Mahalingam (2008), Filippi & al (2009), Rochoux & al (2013) ...



# Empirical wildfire model (McArthur 1966 ...)



### Experimental fires $\rightarrow$ Risk index



Risk index = f (Air T<sup>°</sup> and humidity, Drought factor, Wind speed, Last rainfall)

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Aix\*Marseille Wildfire physics: example of the relationship rate of spread versus wind speed from empirival approach.

## R = A x U<sup>B</sup> With B ranged between 0.5 and 2!

Wind U

### Rate of spread R

Impossibility to transpose from one ecosystem to another one!



# Rate of spread versus wind speed: the impossible extrapolation.



### Aix\*Marseille Université Semi-empirical 1972 Rothermel's model (BEHAVE, FARSITE)





 $\begin{array}{l} \textbf{R} = \xi \ \textbf{Ir} \ / \ (\rho_s \ \alpha_s \ \Delta hi) \ (\textbf{1} + \phi_w + \phi_s), \ \xi = f(\sigma_s) \\ \rho_s \ \alpha_s : \ \text{Fuel density and fuel volume fraction} \\ \textbf{Ir} : \text{Heat of combustion} \\ \Delta hi = C_p \ [T_i - T_a] \ \text{Enthalpy of ignition} \\ \sigma_s : \ \text{Surface area} \ / \ \text{Volume ratio of solid fuel particles} \\ \phi_w \ ; \ \phi_s : \ \text{wind and slope factor} \end{array}$ 

# Malibu fire (22/10/1996) Firetec simulation (LANL)

**AVIRIS Derived Fuel** 



# Malibu fire: semi-empirical model versus physical model

Time necessary to burn 50 ha from the canyon to the top of the hill (real time =10 minutes)

Effect of to evaluate the wind	Slope	Fire	Time
Farsite (Rothermel)	- and the	Sec. 1	180 min.
Hygrad + Rothermel			20 min.
Firetec	X	X	10 min.

### (Aix+Marseille Université Hierarchy of wildfire models from semi-empirical model (top), to coupled meso-scale atmosphericfire (middle), to fully physical model (bottom)



How simulate the behaviour of wildfires using a "fully" physical model ?

Physical phenomena

Compressible flow Reactive flow Turbulent flow Radiation Solid/Gas interface

Solid fuel degradation

### Scales

Wind speed Flame thickness Turbulent structures Extinction length Vegetation Simulating wildfire behaviour using a physical model : multiphase approach



Vegetation Drying Pyrolyse Combustion

### CFD (Navier Stokes) Low Mach

Fire

# Some wildfire fully physical models

	Time integration	Low Mach	Turbulence	TRI model	Combustion model	Multi phase	Lab scale	Large scale
FIRESTAR	Implicite	Yes	LES	Yes	Yes	Yes	Yes	Yes (**)
FDS	Explicite	Yes	LES	No (*)	Yes	Yes	Yes	Yes
FIRETEC	Explicite	No	LES	No (*)	No (***)	Yes	No	Yes
FIREFOAM	Implicite	Yes	LES	Yes	Yes	No	Yes	Yes (**)

(\*) Prescribed global radiant model (\*\*) < 500 m (\*\*\*) Degradation of the vegetation and heat release occur in one single cell

### Aix+Marseille Results obtained for crown fires using Firetec and FDS (ICFME) ...





### Aix\*Marseille Compressible flow: low Mach number approximation (non reactive flow, ideal gas)

**Compressible Navier-Stokes equations**  $\frac{\partial \rho}{\partial t} + \frac{\partial \rho \ U_j}{\partial x_i} = 0$  $\frac{\partial \rho U_i}{\partial t} + \frac{\partial \rho U_j U_i}{\partial x_i} = \frac{\partial \overline{\sigma_{ij}}}{\partial x_i} + \rho f_i$  $\frac{p}{\rho} = \frac{R T}{M}$ 



 $\frac{\Delta p}{p} = O(Mach^2) \text{ Mach} = \frac{U}{a} \quad a: sound speed in air (~340 m s^{-1})$  $U = 50 \text{ km/h} = 14 \text{ m/s} \rightarrow Mach = 0.04 \rightarrow \frac{\Delta p}{p} = 0.0016 \text{ (160 pa)}$  $p = \hat{P} + p' \rightarrow \frac{\hat{P}}{\rho} = \frac{RT}{M} \text{ (in open domain } \hat{P} = \text{cte} = 10^5 \text{ pa)}$ 



# Compressible flow: low Mach number approximation CFL stability criteria (numerical simulation)

 $\frac{V \ \delta t}{\Delta} < C$ Explicit scheme: C < 1 Implicit scheme: C > 1

Full compressible V = max(U, a)



### Low Mach number

V = U

 $\delta t$  : time step  $\Delta$  : mesh size The low Mach number approximation allows theoretically to reduce the time step by a factor equal to 1/Mach

# Some results from the classical homogeneous theory (Kolmogorov 1941)

# $I_{t} \sim H_{Fuel}$ $I_{t}/\eta \sim 3000 - 40000$

 $\leq \pi/1$ 

# $TKE = \int_{0}^{\infty} E(k) \, dk$

E(k)

# Wave number: k

 $2\pi/l_{1}$ 

# Some results from the classical homogeneous theory (Kolmogorov 1941): turbulence modelling

# Simulated using a SGS model $\Delta \sim l_{t} / 6$ $\sim H_{Fuel} / 6$

# $2\pi/l_{+}$ $2\pi/\Delta$ Wave number: k

E(k)

### Aix\*Marseille université Turbulence modelling





### **Turbulence modelling, Reynolds decomposition**

$$U_i = \overline{U_i} + U'_i$$
 with  $\overline{U_i} = \frac{1}{T} \int_t^{t+T} U_i(t) dt$ 

Instantaneous momentum equation (isotherm, incompressible flow)

$$\frac{\partial \rho U_i}{\partial t} + \frac{\partial \rho U_j U_i}{\partial x_j} = \frac{\partial \sigma_{ij}}{\partial x_j} + f_i$$

Average momentum equation

$$\frac{\partial \rho \overline{U_i}}{\partial t} + \frac{\partial \rho \overline{U_j U_i}}{\partial x_j} = \frac{\partial \overline{\sigma_{ij}}}{\partial x_j} + \overline{f_i} - \frac{\partial \rho \overline{U_j U_i}}{\partial x_j}$$

Additional terms  $\rightarrow$  turbulence model

Lot of turbulence models are based on the concept of eddy viscosity, associate to the idea that the turbulent structures are isotropic (assumption only verified for small turbulent structures and not for large scale turbulent structures)

$$\overline{U_j'U_i'} = \frac{2}{3} K \delta_{ij} - \mu_T \left( \frac{\partial \overline{U_i}}{\partial x_j} + \frac{\partial \overline{U_j}}{\partial x_i} \right) \text{ with } \mu_T = \rho L^2 \left| \frac{\partial \overline{U_i}}{\partial x_j} + \frac{\partial \overline{U_j}}{\partial x_i} \right|$$

## Aix\*Marseille **Turbulence model, K-ε model**

Eddy viscosity, effective viscosity

$$\mu_{eff} = \mu + \mu_T \quad \text{with } \mu_T = \rho \ C_\mu \ \frac{\kappa^2}{\epsilon}$$

$$\overline{\sigma_{ij}} = -\left(p + \frac{2}{3} \ K\right) \delta_{ij} + \mu_{eff} \left(\frac{\partial \overline{u_i}}{\partial x_j} + \frac{\partial \overline{u_j}}{\partial x_i}\right)$$

$$\frac{D\rho \overline{u_i}}{Dt} = \frac{\partial \overline{\sigma_{ij}}}{\partial x_j}$$

$$\frac{D\rho K}{Dt} = \frac{\partial}{\partial x_j} \left[\frac{\mu_{eff}}{\sigma_K} \ \frac{\partial K}{\partial x_j}\right] + \rho P - \rho \epsilon$$

$$\frac{D\rho \epsilon}{Dt} = \frac{\partial}{\partial x_j} \left[\frac{\mu_{eff}}{\sigma_\epsilon} \ \frac{\partial \epsilon}{\partial x_j}\right] + \rho \ C_{\epsilon 1} \ \frac{P}{T} - \rho \ (C_{\epsilon 2} + T)$$
T: characteristic time of the turbulence
$$T = \max(\tau, C_T \ \tau_\eta) \ \tau = \frac{\kappa}{\epsilon} \quad \tau_\eta = \left(\frac{\nu}{\epsilon}\right)^{1/2}$$

 $\boldsymbol{\epsilon}$ 



 $R)\frac{\epsilon}{T}$ 

 $\epsilon J$ 

### Aix Marseille Turbulence modeling: how evaluate the turbulence integral length scale I<sub>t</sub>?

**Two sources of production of turbulence:** 

- The shear flow between the vegetation and the boundary layer flow
- The buoyancy between the hot gas in the plume and the ambient air





### E.G. Patton, J.J. Finnigan 2013

#### Aix\*Marseille Universite ABL/canopy interaction (Finnigan 2000)



# ABL/canopy interaction: how evaluating the force induced by the tree upon the air flow?



 $\int \rho\left(\vec{U}.\vec{n}\right)\vec{U}\,dS = \sum \overrightarrow{F_{ex}}$ 

## $\rho(\vec{U},\vec{\nabla})\vec{U} = \vec{F_V} + \text{Div}\,\bar{\bar{\sigma}} - \rho C_D \,LAD \,\|U\|\,\vec{U}$

Integrating the stress forces along the interface air/tree? Problem: the fractal nature of this interface Solution global balance + considering the tree (leaves, twigs, branches...) as a sparce porous media Drag force with a drag coefficient defined from the Leaf Area Density (LAD)

#### Aix+Marseille Universite ABL/canopy interaction



Leaf Area Density (LAD)

$$\frac{\alpha_k \sigma_k}{2} = \text{LAD}$$

$$\frac{D\rho U}{Dt} = \text{Div} \,\overline{\sigma} - \sum_k \rho \, C_D \, \frac{\alpha_k \sigma_k}{2} \|U\| \, U$$
Drag force

#### Leaf Area Density (m<sup>2</sup>/m<sup>3</sup>) Similitude parameter LAI (Leaf Area Index)







#### (Aix\*Marseille Universite Turbulence model, ABL/canopy interaction

Leaf Area Density (LAD)

$$\frac{\alpha_k \sigma_k}{2} = \text{LAD}$$

$$\frac{\partial \rho U}{\partial t} = \text{Div} \,\overline{\sigma} - \sum_k \rho C_k \frac{\alpha_k \sigma_k}{2}$$

$$\frac{D\rho U}{Dt} = \text{Div}\,\overline{\overline{\sigma}} - \sum_{k} \rho \,C_{D} \frac{\alpha_{k} \,\sigma_{k}}{2} \|U\| \,U$$

$$\frac{D\rho K}{Dt} = \frac{\partial}{\partial x_j} \left[ \frac{\mu_{eff}}{\sigma_K} \frac{\partial K}{\partial x_j} \right] + \rho P - \rho \epsilon + \sum_k \rho C_D \frac{\alpha_k \sigma_k}{2} [U^3 - 4UK]$$

$$\frac{D\rho\epsilon}{Dt} = \frac{\partial}{\partial x_j} \left[ \frac{\mu_{eff}}{\sigma_{\epsilon}} \frac{\partial\epsilon}{\partial x_j} \right] + \rho C_{\epsilon 1} \frac{P}{T} - \rho (C_{\epsilon 2} + R) \frac{\epsilon}{T} + \sum_k \rho C_D \frac{\alpha_k \sigma_k}{2} \left[ \frac{3\epsilon}{2 K} U^3 - 6U\epsilon \right]$$

$$T = \max(\tau, C_T \tau_\eta) \tau = \frac{K}{\epsilon} \tau_\eta = \left(\frac{\nu}{\epsilon}\right)^{1/2} \text{(Integral and Kolmogorov time scale } C_T = 6)$$

Additionnal terms resulting from the flow/vegetation interaction (micro-wake) ! C<sub>d</sub> ~ 0.1-0.4 (typical value)



(C<sub>D</sub> defined using LAD (Leaf Area Density) as a reference surface)







• If ad < 0.01 <  $C_D$  > =  $C_D$  ( $R_e$ ) (~ single particule)

- If  $ad > 0.01 < C_D > = f(ad)$  (wake interaction)
- Typical value: ad ~  $\alpha_s$  ~ 10<sup>-3</sup>- 10<sup>-2</sup>  $C_D = 0.38$ (Water Resources Research Vol.35(2) pp.479-489 (1999), H.M. Nepf)

### Aix\*Marseille Université Effect of atmospheric stratification upon fire dynamics.



 $R_i = \frac{N^2}{\left(\frac{du}{dz}\right)^2}$ 

Effect of atmospheric stratification (stability) upon the fire dynamics, thermal plume, aerosols transport, turbulence ... N: Brunt-Väisälä frequency

$$N^2 = \frac{g}{\rho_\theta} \left| \frac{d\rho_\theta}{dz} \right|$$

$$\boldsymbol{\rho}_{\boldsymbol{\theta}} = \boldsymbol{\rho} \left(\frac{\boldsymbol{p}}{\boldsymbol{p}_{0}}\right)^{\boldsymbol{\gamma}}$$

# Weak gravity flow (Kolmogorov): Ri <<1

# $I_{t} \sim H_{Fuel}$ $I_{t}/\eta \sim 3000$ **E(k)** - 40000 $N^2$ R; $\left(\frac{du}{dz}\right)$ $TKE = \int_{0}^{\infty} E(k) \, dk$

Wave number: k

# Moderate gravity flow (Bolgiano-Obukhov): Ri ~1

k-11/5

 $\int^{\infty} TKE = \int^{\infty} E(k) \, dk$ 

**E(k)** 

# Wave number: k

5/3
# Strong gravity flow: Ri >>1 (Bolgiano scaling) $k_{\rm B} \sim (\beta g)^{3/2} \epsilon^{-5/4} \epsilon_{\rm T}^{3/4}$

**E(k)** 

### k<sub>в</sub> ~ 1/500 km<sup>-1</sup> (atmosphere)

### Wave number: k

#### Aix Marseille data, GASP) GASP: Global Atmospheric Sampling Program wavenumber (radians m-1) (NASA) $10^{-5}$ $10^{-4}$ $10^{-3}$ $10^{-2}$ 108 107 meridional zonal wind wind 106 1-5/ -5/3 spectral density (m3s-2) 105 $10^{4}$ $10^{3}$ 95% confidence interval $10^{2}$ 101 $10^{3}$ $10^{2}$ $10^{1}$ $10^{0}$ $10^{-1}$ $10^{4}$ wavelength (km)



#### Transport equation of chemical species ( $\alpha$ )

For each species 
$$\alpha$$
:  
 $\frac{\partial}{\partial t}(g_{\alpha}\rho_{\alpha})+\frac{\partial}{\partial x_{j}}(g_{\alpha}\rho_{\alpha}u_{j}^{\alpha})=\dot{\omega}_{\alpha}$   
Average density and velocity:  
 $\rho = \sum_{\alpha} g_{\alpha}\rho_{\alpha}$   $\rho u_{j} = \sum_{\alpha} g_{\alpha}\rho_{\alpha}u_{j}^{\alpha}$   
Mass fraction:  
 $\rho Y_{\alpha} = g_{\alpha}\rho_{\alpha}$   
 $\frac{\partial}{\partial t}(\rho Y_{\alpha})+\frac{\partial}{\partial x_{j}}(\rho Y_{\alpha}u_{j})=\dot{\omega}_{\alpha}-\frac{\partial}{\partial x_{j}}(\rho Y_{\alpha}(u_{j}^{\alpha}-u_{j}))$   
Diffusion velocity:  
 $-Y_{\alpha}(u_{j}^{\alpha}-u_{j})=D_{\alpha}\frac{\partial Y_{\alpha}}{\partial x_{j}}$   
 $\frac{\partial}{\partial t}(\rho Y_{\alpha})+\frac{\partial}{\partial x_{j}}(\rho Y_{\alpha}u_{j})=\frac{\partial}{\partial x_{j}}(\rho D_{\alpha}\frac{\partial Y_{\alpha}}{\partial x_{j}})+\dot{\omega}_{\alpha}$ 

#### Laminar/turbulent flame

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- Marmy Marm

**Velocity or temperature signals** 

**Transitional Flow** 

Laminar Flow

#### **Combustion, Arrhenius law, turbulence**



Fire: Turbulent flame Soot particules Chemistry Radiation Coupling combustion/turbulence radiation/turbulence

 $Fuel + \nu O_2 \rightarrow (1 + \nu) Produits$ 

Reaction rate:

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 $\frac{\dot{\omega}_{Fuel}}{\dot{\omega}_{Fuel}} = f(Y_{Fuel}, Y_{O2}, T) = -k_0 Y_{Fuel}^a Y_{O2}^b T^c exp\left(-\frac{E}{RT}\right)$  $\frac{\dot{\omega}_{Fuel}}{\dot{\omega}_{Fuel}} = f(\overline{Y_{Fuel}}, \overline{Y_{O2}}, \overline{T}) ? \text{ No because } \frac{\sqrt{T'^2}}{\overline{T}} \gg 1$ 

Aix\*Marseille Average reaction rate: example of a point located in the intermitent zone (50%: T = 500 K; 50%: T = 2000 K)  $k(T) = k_0 \exp\left(-\frac{E}{RT}\right) \frac{E}{T} = 20\ 000K\ (typical\ value)$  $PDF(T) = \frac{1}{2} \left( \delta(T - T_1) + \delta(T - T_2) \right)$  $T_1 = 500$  K and  $T_2 = 2000$  K  $\overline{k}(T) = \int PDF(T) \ k(T) \ dT = \frac{1}{2} \ k_0 \left( exp\left(-\frac{E}{R \ T_1}\right) + exp\left(-\frac{E}{R \ T_2}\right) \right)$  $k(\bar{T}) = k_0 \exp\left(-\frac{2E}{R(T_1 + T_2)}\right)$  $\frac{k(\bar{T})}{\bar{k}(T)} = 0.005$ 

 $\overline{\dot{\omega}}_{Fuel} \neq f(\overline{T}, \overline{Y_{Fuel}}, \overline{Y_{O2}})$ 





# Turbulent flame: the example of a stabilized premixed flame





U'/S<sub>L</sub>: 0.85







Reaction rate (EDC type model):  $\overline{\dot{\omega}_{Fuel}} = \frac{\overline{\rho}}{\tau} \min\left(\widetilde{Y_{Fuel}}, \frac{\widetilde{Y_{02}}}{\nu}\right) \quad (kg \ m^{-3}s^{-1})$ 



#### **Temperature fluctuations in a turbulent flame**



Figure 1. Superposition of a time-averaged photograph of the flame (1/50th sec) and an instantaneous ( $10^{-4}$  sec) flame contour obtained from a tomographic cut. The ring of pilot flames is visible. Mean flow velocity : 12 m/s, Equivalence ratio : 0.75.



#### How describing a vegetation layer ?



**Physical properties** 

• Density

- Volume fraction
- Surface Area/Volume (SA/V)
- Fuel moisture content (FMC)



#### Solid fuel classification



#### Equilibrium time (FMC/Air):

- 1H φ : 0-0.64 cm
- 10H φ : 0.64-2.54 cm
- 100H φ : 2.54-7.62 cm



#### **Fire residence time**





## Thermal analysis of Mediteranean vegetation samples (INRA-Avignon)











#### (Aix\*Marseille Université Initiative d'excellence Transport equations in the gas phase<sup>3</sup>



Aix+Marseille Radiation transfer equation + TRI Optically Thin Fluctuation Approximation (OTFA) P.J. Coelho Prog. Energ. & Combust. Science (2007) (neglecting the scaterring)

$$\frac{dI_{\nu}}{ds} = K_{\nu}(I_{B\nu} - I_{\nu})$$
$$\frac{d\bar{I}_{\nu}}{ds} \approx K_{\nu}(\bar{T})I_{B\nu}(\bar{T}) \left[1 + 6 \frac{\bar{T'}^2}{\bar{T}^2} + 4 \frac{\bar{T'}^2}{\bar{K}\bar{T}} \frac{\partial K_{\nu}}{\partial T}\Big|_{\bar{T}}\right]$$
$$-K_{\nu}(\bar{T}) \bar{I}_{\nu}$$

OTFA theoretical domain of validity:  $K_{\nu} \times l_t \ll 1$ 

$$\frac{(\bar{I'^2})^{1/2}}{\bar{I}} \approx 20\% \ to \ 500\%$$

Aix Marseille Radiation transfer equation + TRI Optically Thin Fluctuation Approximation (OTFA) P.J. Coelho Prog. Energ. & Combust. Science (2007)

$$\frac{d\alpha_{g}\overline{I}}{ds} = \alpha_{g}\left(\frac{\sigma KT^{4}}{\pi} - \overline{\sigma_{a}}\overline{I}\right) + \sum_{i}\left[\frac{LAD}{2}\left(\frac{\sigma T_{s}^{4}}{\pi} - \overline{I}\right)\right]$$
$$\overline{KT^{4}} \approx \overline{K}\overline{T}^{4}\left[1 + 6\frac{\overline{T'^{2}}}{\overline{T}^{2}} + 4\frac{\overline{T'^{2}}}{\overline{K}\overline{T}}\frac{\partial K}{\partial T}\Big|_{\overline{T}}\right]$$

 $K = K_{Pro} + K_{Soot} = 0.1 X_{Pro} + 1862 f_v T$ 

### FireStar3D Model

#### Fluid Mixture

> Low Mach number formulation (Navier-Stokes equations)

- ➤ Turbulence: k-ε and LES approaches
- Heat Transfer: Enthalpy formulation + Radiation (RTE)
- Species: Transport + Combustion in fluid phase (EDC model)
- Soot: Transport equation + Oxidation

#### Solid Particles

Drying, Pyrolysis & Combustion models
 Mass, energy and particle size balances

Fluid Mixture/Solid coupling

- Aerodynamics (porous media)
- Heat transfer
- Species exchange

#### **Solid Fuel Combustion Model**



#### **Fluid-Phase Model**

 $\circ \quad \frac{D\overline{\rho}}{Dt} = \sum \sum \dot{M}_{\alpha}^{m}$  $\circ \frac{D(\overline{\rho u_i})}{Dt} = -\frac{\partial \overline{P}}{\partial x_i} + \frac{\partial}{\partial x_i} \left[ \overline{\mu} \left( \frac{\partial \overline{u_i}}{\partial x_i} + \frac{\partial \overline{u_j}}{\partial x_i} - \frac{2}{3} \frac{\partial \overline{u_l}}{\partial x_i} \right) - \frac{\partial}{\partial x_i} \left( \overline{\rho u_i' u_j'} \right) + \left( \overline{\rho} - \overline{\rho}_0 \right) g_i - \sum_m F_{Di}^m$  $\circ \frac{D(\overline{\rho}\overline{h})}{Dt} = \frac{\partial}{\partial x_{j}} \left( \frac{\overline{\mu}}{Pr} \frac{\partial \overline{T}}{\partial x_{j}} \right) - \frac{\partial}{\partial x_{j}} \left( \overline{\rho}u'_{j}h' \right) - \frac{d\overline{P}_{th}}{dt} + \left( 1 - \alpha_{sG} \right) \Delta h_{Char} \sum_{m} \dot{\omega}_{Char}^{m}$  $+\sum \sum \dot{M}_{\alpha}^{m} \overline{h}_{\alpha}^{m} - \sum \dot{Q}_{S,Conv}^{m} + \alpha_{G} \sigma_{G} \left( J - 4\sigma \overline{T}^{4} \right)$  $\circ \frac{D(\overline{\rho}\overline{Y}_{\alpha})}{Dt} = \frac{\partial}{\partial x_{i}} \left( \frac{\overline{\mu}}{Sc} \frac{\partial \overline{Y}_{\alpha}}{\partial x_{i}} \right) - \frac{\partial}{\partial x_{i}} \left( \overline{\rho}u_{j}'Y_{\alpha}' \right) + \overline{\omega}_{\alpha} + \sum_{m} \dot{M}_{\alpha}^{m}$ 

• Ideal Gas Equation

Morvan *et al.* (2001, 2004, 2008, 2009, 2018)

#### **Numerical Approach**

#### □ Formulation

- Fully Implicit (Non-conditional temporal stability)
- Segregated Formulation (PISO algorithm)

#### Numerical Method

Finite Volume Method on Cartesian non-uniform grid
 ~ 3<sup>rd</sup> order space accuracy (QUICK scheme)
 > 3<sup>rd</sup> order time precision with Adaptive time-stepping
 Radiation: Discrete Ordinate Method (DOM)- S8
 Parallel Computing: OpenMP directives

#### **Fluid-Phase and Solid-Phase Meshes**





### **Firestar3D: Boundary conditions**

- **Dirichlet** (imposed  $\phi$  profiles) or Neumann  $\left(\frac{d\phi}{dn} = const.\right)$
- **Periodic conditions in any direction**
- **Reflective conditions for the RTE**
- Time-dependent BC may be customized





### Firestar3D: Fire can creates its own wind

- A pressure gradient is applied to impose  $U_{top}$   $\frac{dP^*}{dx} = \frac{\rho U^*}{dt} \left(1 \frac{U^*}{U_{top}}\right)$
- The code determines velocity distribution at inlet
- Once  $U_{top}$  is reached,  $\frac{dP}{dx} \rightarrow 0$



#### Experimental fire in shrubland (EU Firestar project, Galicia-Spain)



#### Aix\*Marseille Experimental fire in shrubland (EU Firestar project, Galicia-Spain)



#### Fuel: Ulex (Europaeus, Minor) (ajoncs)

- Fuel Families = 14
- FMC:

108-150 % (living), 10-32% (dead)

- Fuel depth = 1.25 m,
- Wind: 5.7 m/s (z=10 m),
- Slope : 5°





#### Experimental fire (EU Firestar project, Galicia-Spain)



- Experiment: ROS = 0.273 m/s
- Simulation : ROS = 0.248 m/s







Consequence: in some situations the wind flow can cross the fire front, affecting a lot the wind/fire interaction, this event cannot be reproduced in 2D !





#### Numerical simulations of grassland fires





### Grassland fire: rate of spread (ROS) vs ignition line width (w) (border effect)



#### **Cheney and Gould IJWF 1995**



## Numerical simulations of grassland fires (with periodic lateral boundary conditions)





#### Numerical simulation of grassland fire (with periodic BC) U = 1 m/s





#### Numerical simulation of grassland fire (with periodic BC) U = 10 m/s


#### Flames Patterns ↔ Wind Speed

U = 1 m/s

U = 5 m/s



### Wavelength of Coherent Structures (Plumes)

U = 5 m/s



## Wavelength of Coherent Structures (Plumes)



$$F_{r} = \frac{g \lambda}{(U_{10} - ROS)^{2}}$$
$$N_{C} = \frac{2 g I}{\varrho_{0} C_{p0} T_{0} (U_{10} - ROS)^{3}}$$

#### Exp. Data: Finney et al., 2013



#### Aix\*Marseille Université Fire safety engineering applied to wildfire: how design a fuel break ?









#### How design a fuel break: some empirical formula





.....

### Some critical heat fluxes

Exposure without risk (skin)	1 kW/m <sup>2</sup>
Firefighter	7 kW/m <sup>2</sup>
Skin, 3 s exposure (pain)	10.4 kW/m <sup>2</sup>
Skin 5 s exposure (2 <sup>nd</sup> degree burning)	16 kW/m <sup>2</sup>
Wood 60 s exposure (ignition)	31 kW/m <sup>2</sup>



# With a fuel break : $U_1 = 8$ m/s, w = 0.5 kg/m<sup>2</sup>, Nc = 4.2, $L_{FB} = 20$ , 19 et 10 m





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#### Optimal fuel break width vs Byram's convective number (N<sub>c</sub>)

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# **Conclusions (I)**

- As a complementary approach to experimental fires CFD modeling can contribute in the understanding of wildfires behaviour.
- CFD wildfire modeling regroups various level of description in order to reproduce fires at different scales (from < 500 m to >> 1km)
- At a quite large scale, a simple coupling between an atmospheric mesoscale model with a simplied fire propagation model is sufficient to forecast the trajectory of the thermal plume associated to the development of a wildfire (air quality, airport activity ...).
- At a more local scale, a detailed CFD approach can contribute to the understanding of wildfires behaviour, especially the effects of slope, wind and other physical parameters upon the fire front dynamic, the identification of regimes of propagation (wind driven, plume dominated, slope driven ...) ...
- As a postfire or a fire safety analysis CFD tools can contribute to understand some critical phenomena occuring during fire fighting operations or help to the design of fire safety layout (fuelbreak ...)



# **Conclusions (II)**

- Correctly define the objectives of the numerical study,
- Don't forget the state of the art of CFD modeling (mesh design, numerical scheme, convergence monitoring ...),
- Computational effeciency needs also by a great attention on physical models (low Mach number approximation, turbulence and combustion modeling, heat transfer modeling, boundary conditions, size of the computational domain ...)
- Don't limit the analysis to few numerical results, exploring the trends ...
- As possible, comparing with experimental data,
- Don't forget that a model is not the full reality, keep in mind the limitations, no model can be considered as validated (this qualification must be bannied to all scientific report).
- To generalize and compare numerical/experimental results, dimensional analysis can be a good option (Byram's convective number, Froude number, Leaf Area Index ...)
- Round trip between CFD fire models and simplified physical models can be very often useful.









 $\frac{\partial \rho \phi}{\partial t} = F(t, \phi...)$ Euler explicit:  $\frac{(\rho\phi)^{n+1}-(\rho\phi)^n}{2}=F(\phi...)|_n$ Euler implicit:  $\frac{(\rho\phi)^{n+1} - (\rho\phi)^n}{F(\phi...)_{n+1}} = F(\phi...)_{n+1}$ 

Stability criterium  $CFL = rac{U \,\delta t}{\delta h} < 1$  $CFL = \frac{U \, \delta t}{\delta h} < C_{Max}$  $C_{Max} > 1$ 



Exact equation:

 $\rho u \frac{\partial \phi}{\partial r} = \Gamma \frac{\partial^2 \phi}{\partial r^2}$ 

Equation modélisée :

Convective part: Second order scheme:

$$\rho u \frac{\partial \phi}{\partial x} = \Gamma \frac{\partial^2 \phi}{\partial x^2} + O(\delta h^2)$$

 $\rho u \frac{\partial \phi}{\partial x} = \left( \Gamma + \frac{\rho u \,\delta h}{2} \right) \frac{\partial^2 \phi}{\partial x^2} + O(\delta h^2)$ 

First order scheme:

 $P_e = \rho \frac{u \,\delta h}{\Gamma} \,(\text{Péclet mesh number})$ 



Second order centred scheme:

$$\rho u \left( \frac{\phi_{i+1} - \phi_{i-1}}{2\delta h} \right) = \Gamma \frac{\phi_{i+1} + \phi_{i-1} - 2\phi_i}{\delta h^2}$$

First order upwind scheme:

$$\rho u \left( \frac{\phi_i - \phi_{i-1}}{\delta h} \right) = \Gamma \frac{\phi_{i+1} + \phi_{i-1} - 2\phi_i}{\delta h^2}$$





$$\frac{\phi_{i+1} - \phi_{i-1}}{2\delta h} = \frac{\partial \phi}{\partial x}\Big|_{i} + \frac{\partial^{3} \phi}{\partial x^{3}}\Big|_{i} \frac{\delta h^{2}}{2} + \dots$$
$$\frac{\phi_{i} - \phi_{i-1}}{\delta h} = \frac{\partial \phi}{\partial x}\Big|_{i} - \frac{\partial^{2} \phi}{\partial x^{2}}\Big|_{i} \frac{\delta h}{2} + \dots$$
$$\frac{\phi_{i+1} + \phi_{i-1} - 2\phi_{i}}{\delta h^{2}} = \frac{\partial^{2} \phi}{\partial x^{2}}\Big|_{i} + \frac{\partial^{4} \phi}{\partial x^{4}}\Big|_{i} \frac{\delta h^{2}}{12} + \dots$$



Schéma centré du 2nd ordre:

$$\rho u \frac{\partial \phi}{\partial x} = \Gamma \frac{\partial^2 \phi}{\partial x^2} + O(\delta h^2)$$

Schéma décentré du 1er ordre:

$$\rho u \frac{\partial \phi}{\partial x} = \left(\Gamma + \frac{\rho u \,\delta h}{2}\right) \frac{\partial^2 \phi}{\partial x^2} + O(\delta h^2)$$